INTERAGENCY SECURITY CLASSIFICATION APPEALS PANEL，
E．O．13526，SECTION 5．3（b）（3）
ISCAP APPEAL NO．2009－068，document no． 79
DECLASSIFICATION DATE：December 5， 2014

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（ 4.4 EXEMPTED FROM DECLASSIEICATION IAW EO 12958 REVIEW DATETVM Y7 TREULEWER REFER TO A．\＆4NT）
EXPMIIION（S）：（1）2 3．4．5．6．7．8．9
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SEPTEMBER 2009

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WIR 23165 2．Pe4 Jun 1965



One vehicle--Zond 2, a Mars prope - = has been dropped since publication of the last list (WIR. $19 / 65$ ), and one $=$ C Cosmos $67-2$ has been added. The latter will probably be de-orbled betore this WIR is distributed.

All the above intercepts Vere transmitted by yehicies launched this year; a11 Soviet pay loads launched in previous years and stilliniorbit apparently have ceased transmitting.
(Various EINT sensors)
TSECRET NO FOREICNDISSEMINATION--Releasable to US: UK \& Canada)

## New Propulsion Combination Used for Triple-Payload Launch of 21 February

The Soviets to date have used their first ICBM, the SS-6, for almost all of their space launches from the Tyuratam Missile Test Range. The rnost recent exception, system used for the 21 February launch of Cosmoses 54, 55 \& 56 aud th 15. Narch launch of Cosmoses 61,62 \& 63. Although ceived during the first Soviet triple-payload launch (Cosmoses 38, 39 \& 40) of 18 August 1964 , it appears reasonable to assume that the same propulsion system was used on that occasion.
4) The new vehicle is of tandem, 2-stage design. The first stage more closely resembles the SKEAN/SS-5 IRBM than any other knovn Soviet Missile. The second stage appears to be an essentially new stage, able to restart in orbit. This new propulsion systen, FTD estimates, could put a useful payload of about 3500 pounds into a $100-n$. m. circular orbit, Actual. paylaads in orbit probably totaled much less than 3500 pounds, at leastifor the Cosmos 54-55-56 launch.

The new propulsion system was probably devised to be used with a new class of utilitarian Soviet satellites in the intermediate weight range -for communications, weathex, oceanographic, navigation, or geodetic purposes -- whichiwill probably appear in increasing numbers in coming months, Heretofore, most Soviet space payloads have either been heavyweights in the 10,000 to 15,000 -pound weight range or relatively simall vehicles in the 300- to 500 - pound range. The heavywights include the manned Vostoks and Voskhods, the recoverable biosatelites of 1960 and 1961 , the xecoverable eosmos photorecce satellites launched from Thuratam, and the parkingorbit platforms for injection of tunar and intexplanetary probes. The lightweights have been, primarily; the research-iype. Cosmoses launched from Kapustan Yar ( $(X X)$. The heayyweights were all launched by the SS -6 ; the lightweights by a relatively small 2 -stage vehicle. The SS- 6 would be uneconomical to use with the new uthitacian satellites, the KY vehicle would be inadeguate.

The restartablewpper stage of the new propulsion systerm vould also appearto be specially suited to utilita xian satellites, affording possibly the
24 capability to separate mulliple payloads at desiredintervals in space ont to make adjustments in orbital parameters.
(FTD: NORAD)
(SFCDEE NO FOREIGN DSSFMMATION Cosmos 67 De-Orbited Routinely on Rev 128
21. Cosmos 67 , which the Soviets launched from Tyuratam at about 1050 Z , 25 May, was de-orbited on Revolution 128 and probably impacted in the USSR at alout 1031-10362, z. June 1965, after nearly 8 days in orbit. This is the Sixth photoreconnaissance satellite launched by the Soviets this y ear, and the first this year to haye an orbital inclination to the Equator of 51 degzees. (SPADATS; NORAD)
TSECNFT NO FOREIGN DISSEMINATION -- Releasable to MS, UK \& Canada)

